

Advanced Aircraft Design

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Amsterdam (Netherlands)

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Introduction

Modern aircraft design requires achieving a precise balance between performance, safety, weight, cost, operational efficiency, and environmental impact. A successful aircraft cannot be developed by optimizing the wing, propulsion system, or airframe independently, because changes to any major subsystem directly influence the performance and characteristics of the entire aircraft.

The Advanced Aircraft Design course provides a comprehensive methodology for understanding every stage of the aircraft design process, beginning with mission analysis and the conversion of operational requirements into engineering specifications. The course progresses through conceptual design, preliminary sizing, weight estimation, wing and fuselage configuration, propulsion system selection, and concludes with performance evaluation, stability analysis, safety considerations, and certification requirements.

The program focuses on the conceptual and preliminary design phases, where the most critical engineering decisions are made regarding aircraft configuration, passenger or payload capacity, range, cruise speed, operating altitude, payload capability, and overall structural layout. Participants will learn how engineering calculations, analytical models, simulation tools, and trade-off studies are used to compare alternative concepts and identify the most suitable design for specific operational requirements.

The course is designed for aerospace engineers, aeronautical engineers, mechanical engineers, structural engineers, propulsion specialists, aerodynamicists, flight control engineers, simulation professionals, personnel working within aviation manufacturers, airlines, research and development organizations, and aviation regulatory authorities. It is equally valuable for engineering project managers, design team leaders, technical analysts, and professionals involved in evaluating both manned and unmanned aircraft programs.

By the end of the course, participants will possess a practical framework for understanding the relationship between mission requirements and engineering decisions, evaluating the impact of design modifications, comparing alternative concepts systematically, and developing an integrated preliminary aircraft design that balances performance, safety, manufacturability, and operational efficiency.

Course Objectives

The Advanced Aircraft Design course aims to develop participants' ability to analyze aircraft requirements and transform them into integrated engineering solutions. Upon completion of the course, participants will be able to:

The graphic features the text 'UK Training' in a small font above the word 'PARTNER' in a large, bold, black font. This text is positioned over a background of concentric circles and a chessboard pattern with several chess pieces (a king, a queen, and a pawn) in the foreground.

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- Understand the major phases of the aircraft design process.
- Differentiate between conceptual, preliminary, and detailed aircraft design.
- Convert operational and mission requirements into measurable engineering specifications.
- Select the appropriate aircraft configuration according to mission objectives.
- Develop a complete aircraft mission profile.
- Estimate maximum takeoff weight, fuel weight, payload, and structural weight.
- Evaluate the effect of aircraft weight on performance, range, and fuel consumption.
- Select appropriate wing geometry and aerodynamic characteristics.
- Analyze the relationship between lift, drag, and angle of attack.
- Estimate fundamental aerodynamic performance parameters.
- Determine propulsion requirements throughout all phases of flight.
- Compare conventional, electric, and hybrid propulsion systems.
- Analyze the integration between propulsion, wing, and airframe design.
- Evaluate the primary structural loads acting on an aircraft.
- Select suitable materials based on strength, weight, cost, and fatigue resistance.
- Understand the principles of fuselage, wing, and empennage design.
- Analyze longitudinal, lateral, and directional stability.
- Estimate flight control requirements and control surface sizing.
- Evaluate takeoff, climb, cruise, and landing performance.
- Develop aircraft performance charts and compare design alternatives.
- Apply engineering trade-off studies to support design decisions.
- Integrate safety, maintainability, and manufacturability into aircraft design.
- Understand the relationship between aircraft design and airworthiness certification.
- Evaluate environmental performance, noise, and energy efficiency.
- Develop an integrated preliminary aircraft concept for a defined mission.
- Document engineering assumptions, constraints, and design decisions clearly.

Course Outlines

Day One: Mission Requirements and Conceptual Aircraft Design

- Understanding the stages of aircraft development from concept to certification.
- Analyzing operator, market, and mission requirements.
- Defining payload, range, speed, altitude, and operational objectives.
- Developing a complete mission profile.
- Converting operational requirements into engineering design criteria.
- Selecting the overall aircraft configuration.
- Comparing conventional, high-wing, low-wing, and blended-wing configurations.
- Evaluating engine placement, empennage configuration, and landing gear arrangement.
- Performing preliminary weight estimation and mass distribution.
- Using reference aircraft to support conceptual design decisions.
- Developing a structured trade-off matrix for alternative concepts.
- Practical exercise on developing a conceptual aircraft design.

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Day Two: Aerodynamics, Wing Design, and Aircraft Performance

- Reviewing lift, drag, thrust, and weight relationships.
- Analyzing airfoil characteristics and aerodynamic behavior.
- Selecting wing area, aspect ratio, and planform geometry.
- Determining sweep angle, taper ratio, and geometric twist.
- Evaluating the influence of wing geometry on efficiency and stability.
- Estimating parasite drag and induced drag.
- Developing aerodynamic lift-drag polar curves.
- Analyzing low-speed performance and stall characteristics.
- Determining high-lift device and control surface requirements.
- Estimating takeoff and landing distances.
- Evaluating climb, cruise, endurance, and range performance.
- Comparing wing configurations using engineering calculations and simulations.
- Practical application involving wing selection for specified mission requirements.

Day Three: Propulsion Systems, Weight Analysis, Structures, and Materials

- Determining propulsion requirements throughout the mission profile.
- Comparing jet, turboprop, electric, and hybrid propulsion systems.
- Evaluating thrust-to-weight ratio and energy consumption.
- Integrating propulsion systems with aircraft structures.
- Assessing the effect of engine location on drag, stability, and maintenance.
- Estimating structural, payload, fuel, and systems weight.
- Calculating center of gravity and operational loading limits.
- Identifying maneuver, landing, and turbulence loads.
- Understanding load paths throughout the wing and fuselage.
- Reviewing the design principles of spars, ribs, and aircraft skin structures.
- Comparing aluminum alloys, titanium alloys, and composite materials.
- Evaluating fatigue, corrosion, and damage tolerance.
- Practical exercise on aircraft mass estimation and center of gravity analysis.
- Comparing material alternatives based on performance and lifecycle cost.

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A graphic illustration of a chessboard with several chess pieces. A large gold king piece is prominent in the foreground, with a silver pawn and a gold pawn nearby. The board is checkered, and there are concentric circles in the background.

Day Four: Stability, Flight Control, Aircraft Systems, and Safety

- Analyzing static and dynamic aircraft stability.
- Determining wing and empennage position relative to the center of gravity.
- Estimating horizontal and vertical stabilizer sizing.
- Evaluating pitch, roll, and yaw control.
- Assessing flight control effectiveness throughout all flight phases.
- Understanding the influence of mass distribution on aircraft dynamics.
- Reviewing mechanical and fly-by-wire flight control systems.
- Examining sensors, flight computers, and actuators.
- Integrating electrical, hydraulic, and environmental systems.
- Understanding redundancy, fault tolerance, and system reliability.
- Conducting preliminary aircraft systems safety analysis.
- Evaluating the impact of certification requirements on aircraft design.
- Assessing maintainability and inspection considerations.
- Simulation exercise involving center-of-gravity changes and system failures.

Day Five: Design Integration, Optimization, Certification, and Final Assessment

- Integrating aerodynamic, propulsion, structural, and control system results.
- Verifying consistency between aircraft weight, performance, range, and payload.
- Identifying critical design variables and engineering constraints.
- Performing trade-off studies involving performance, cost, and safety.
- Applying multidisciplinary design optimization techniques.
- Evaluating the impact of modifying one subsystem on the entire aircraft.
- Assessing manufacturability, assembly, and maintenance considerations.
- Integrating fuel efficiency, energy consumption, noise, and emissions objectives.
- Understanding the general principles of aircraft airworthiness certification.
- Reviewing engineering verification and compliance activities.
- Identifying ground testing, prototype testing, and simulation requirements.
- Developing a design assumptions and risk register.
- Completing a comprehensive aircraft design case study.
- Presenting engineering recommendations and design justifications.
- Conducting the final technical assessment and identifying future development opportunities.

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Why Attend this Course: Wins & Losses!

- Develop a comprehensive understanding of the aircraft design process.
- Strengthen the ability to convert mission requirements into engineering solutions.
- Improve aircraft configuration selection skills.
- Understand the interaction between wings, structures, propulsion systems, and flight controls.
- Improve preliminary aircraft weight and performance estimation capabilities.
- Strengthen aerodynamic analysis skills.
- Evaluate different propulsion technologies more effectively.
- Develop practical knowledge of aircraft structures and materials.
- Improve stability and flight control analysis.
- Apply engineering trade-off studies rather than isolated design decisions.
- Integrate safety and maintainability into early design stages.
- Understand the influence of certification requirements on engineering decisions.
- Use engineering simulations and data to compare alternative concepts.
- Improve engineering documentation and technical reporting skills.
- Complete a multidisciplinary aircraft design study.

Conclusion

The Advanced Aircraft Design course provides a structured methodology for understanding aircraft development as a multidisciplinary engineering process. It integrates mission analysis, requirements definition, configuration selection, aerodynamic design, weight estimation, propulsion selection, structural analysis, stability assessment, performance evaluation, and safety considerations into one cohesive design framework.

Aircraft design begins by defining the mission the aircraft must perform. Mission requirements include passenger capacity or payload, range, cruise speed, operating altitude, runway length, environmental conditions, and safety objectives. These requirements establish the engineering criteria upon which every subsequent design decision is based.

Once mission requirements are established, the design process progresses to selecting the overall aircraft configuration. Decisions include wing location, empennage arrangement, engine placement, fuselage geometry, and landing gear configuration. No single configuration is ideal for every application, since regional transport aircraft, cargo aircraft, surveillance platforms, and training aircraft all require different design priorities.

The course examines aerodynamics as the foundation of aircraft performance. Wing design directly influences lift, drag, speed, stall behavior, range, and energy consumption. Selecting wing area, aspect ratio, airfoil geometry, and sweep angle therefore requires careful analysis of mission objectives and operational constraints.

Participants also examine the relationship between propulsion systems and aircraft design. Engine selection involves much more than thrust output. Engineers must consider aircraft weight, fuel or energy consumption, acoustic performance, cooling requirements, maintenance accessibility, and structural integration. These considerations become even more complex when evaluating different propulsion technologies.

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Weight estimation and center of gravity analysis remain critical throughout aircraft development. Aircraft weight influences takeoff performance, climb capability, cruise efficiency, range, and landing performance, while center of gravity directly affects aircraft stability and controllability. Weight estimates must therefore be continuously updated as the design evolves.

The course further explores aircraft structures and material selection. Engineers must design structures capable of resisting maneuver, landing, and turbulence loads while minimizing structural weight. Material selection also requires consideration of fatigue life, corrosion resistance, damage tolerance, inspection requirements, and lifecycle cost.

Aircraft stability and controllability are equally important design objectives. Aircraft must remain stable throughout all operating conditions while responding predictably to pilot commands or flight control systems. Achieving this balance requires careful sizing and positioning of stabilizers and flight control surfaces.

Safety and certification requirements are integrated throughout the design process rather than addressed at the end of development. Airworthiness regulations influence structural design, aircraft systems, fault tolerance, testing procedures, documentation, and verification activities from the earliest design stages.

The course concludes by emphasizing multidisciplinary design integration. Reducing wing area may decrease aircraft weight but increase takeoff and landing speeds. Increasing engine thrust may improve climb performance while increasing structural weight and energy consumption. Engineering trade-off studies provide the analytical framework required to evaluate these competing objectives and achieve balanced design solutions.

By the end of the course, participants will possess a structured methodology for transforming mission requirements into preliminary aircraft concepts, analyzing aerodynamic performance, propulsion systems, structural behavior, aircraft stability, and engineering trade-offs. This knowledge provides a solid foundation for advanced aircraft development, simulation, testing, and future engineering design activities.

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